



University  
of Glasgow

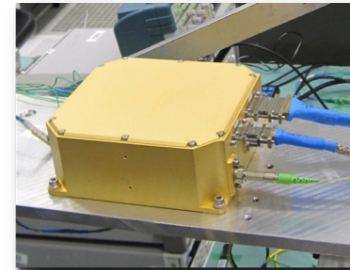
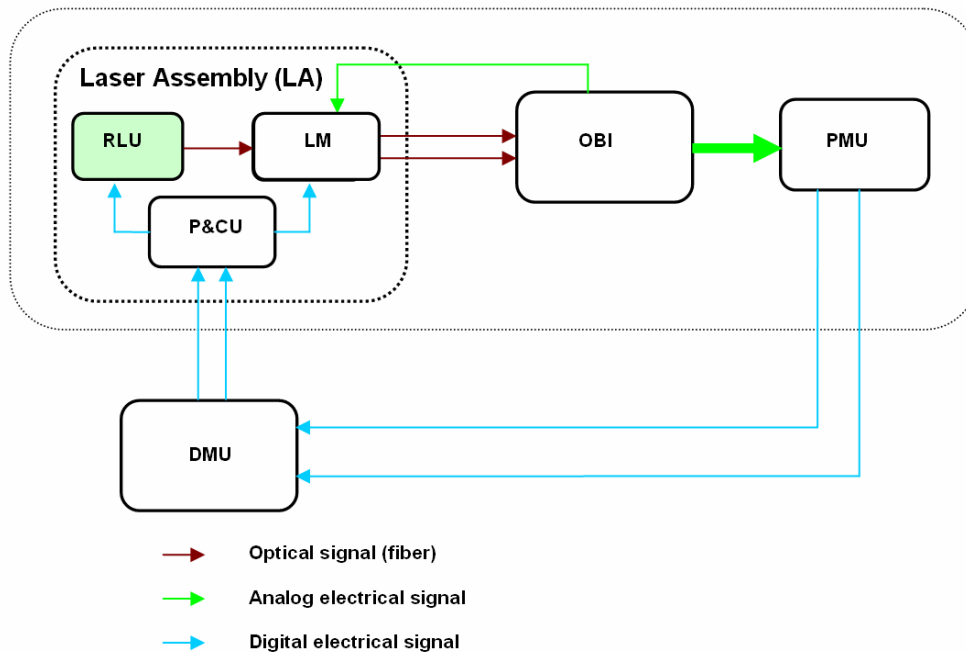


## Outline

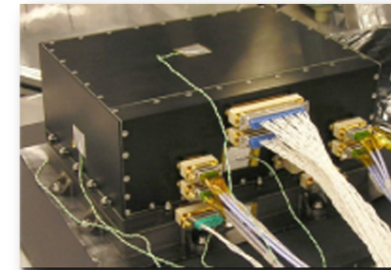
- LTP Optical Metrology System
  - Component parts
  - Main functions
- Requirements and constraints on its performance
  - Source of requirements
  - How they are met
    - Experimental results where possible
    - Modelling where direct ground testing impossible
    - Unexpected features and their solutions
- Summary of how well it performs

## Optical Metrology System (OMS)

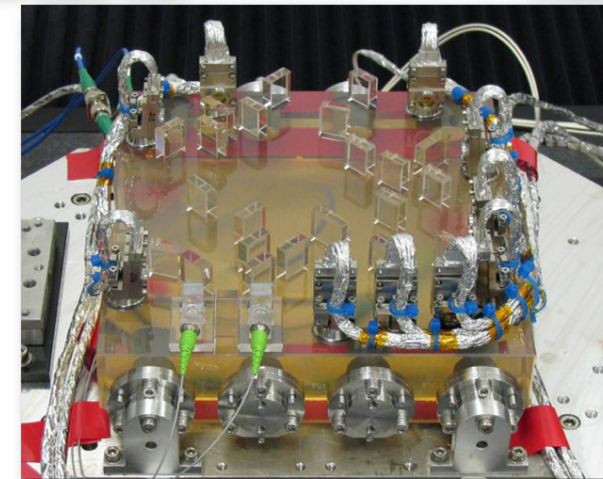
- Reference Laser Unit (RLU)
- Laser Modulator Unit (LM)
- Optical Bench Interferometer (OBI)
- PhaseMeter Unit (PMU)
- Data Management Unit (DMU)



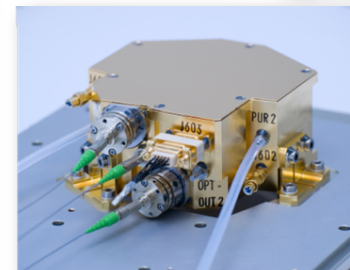
Reference Laser Unit



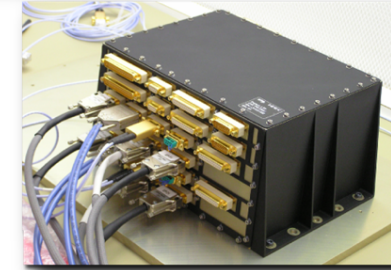
PhaseMeter



Optical Bench Interferometer



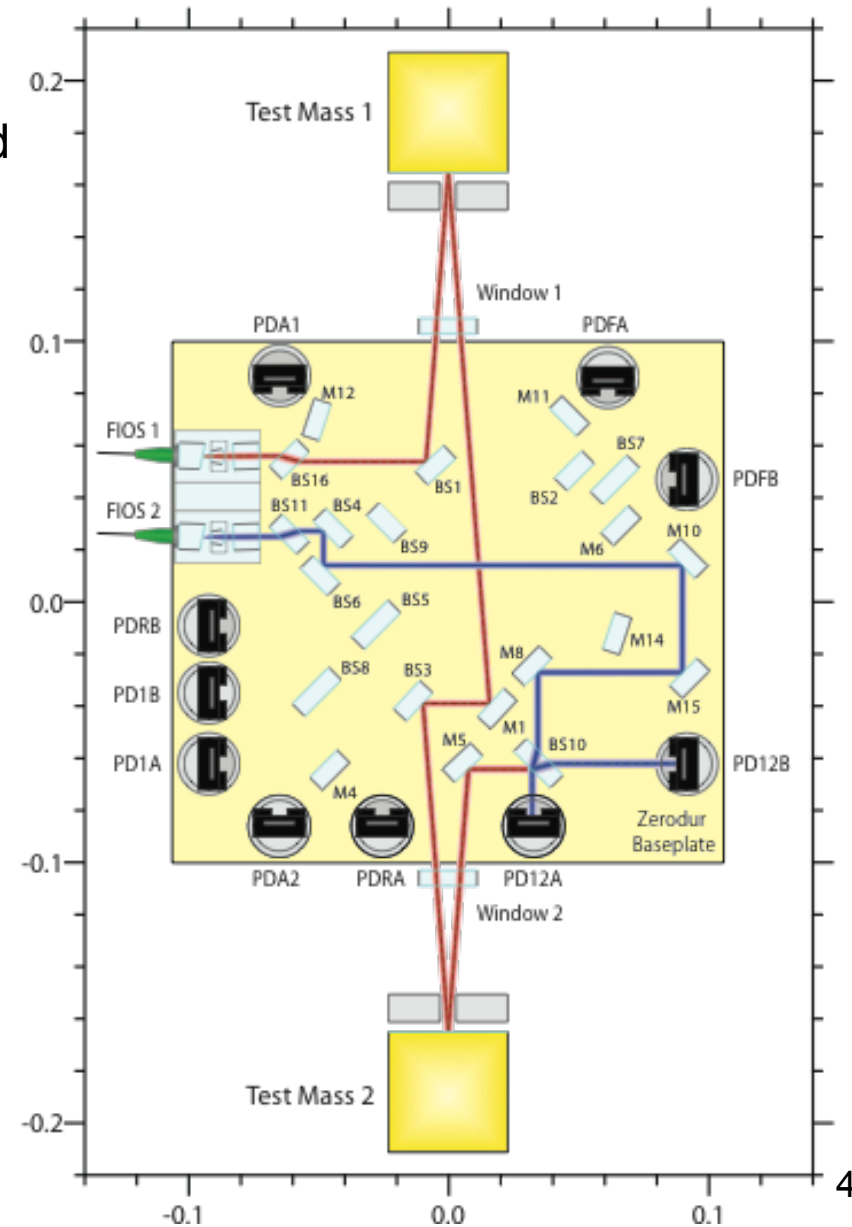
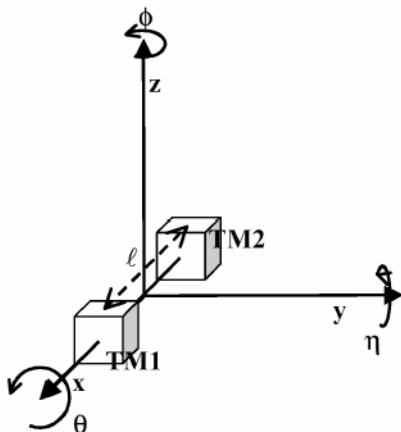
Laser Modulator



Data Management Unit

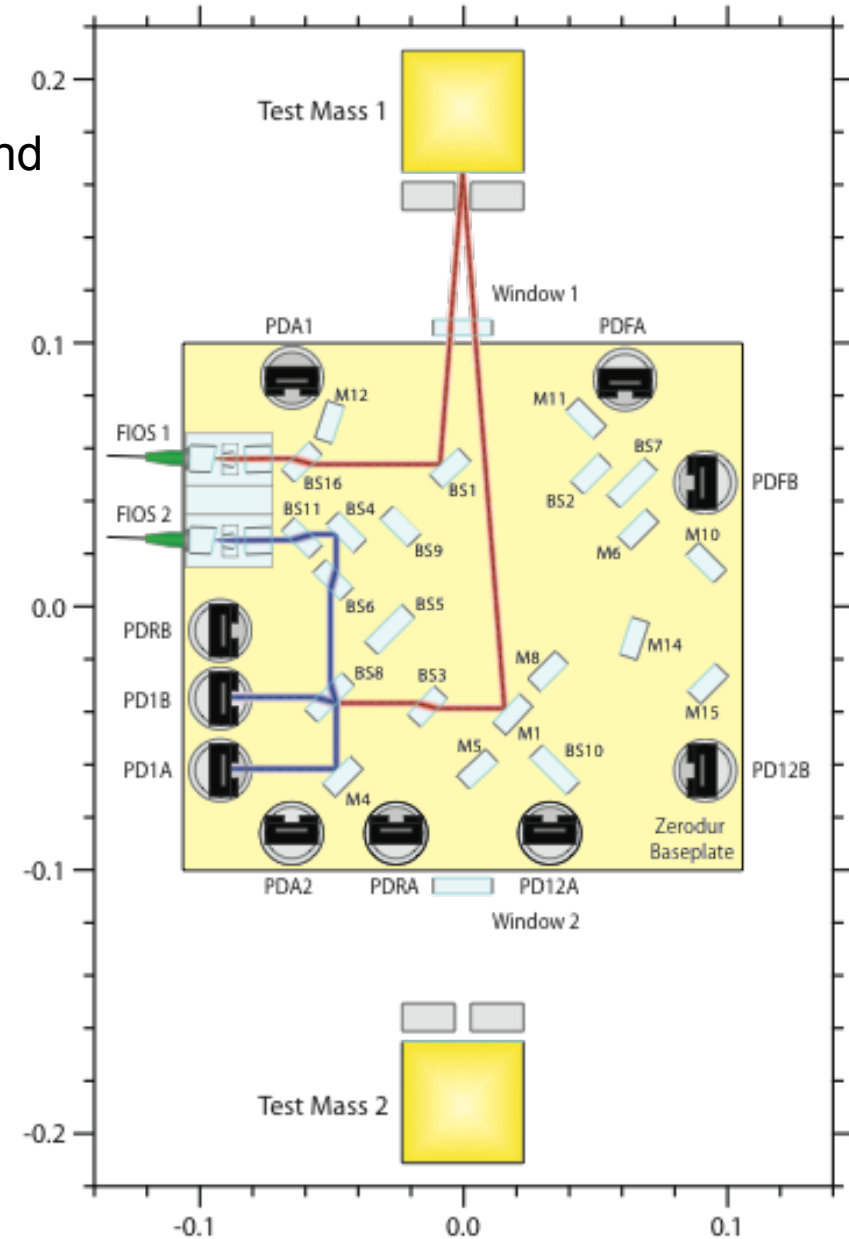
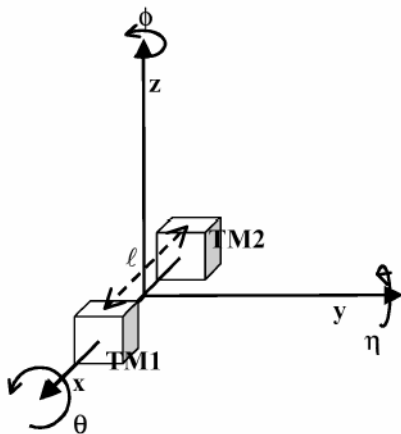
## X12 interferometer

- Relative displacement between test mass 1 and test mass 2
  - $X_{12}$
- Relative angles between test mass 1 and test mass 2
  - Differential Wavefront Sensing (DWS)
  - $\eta_{12}, \phi_{12}$



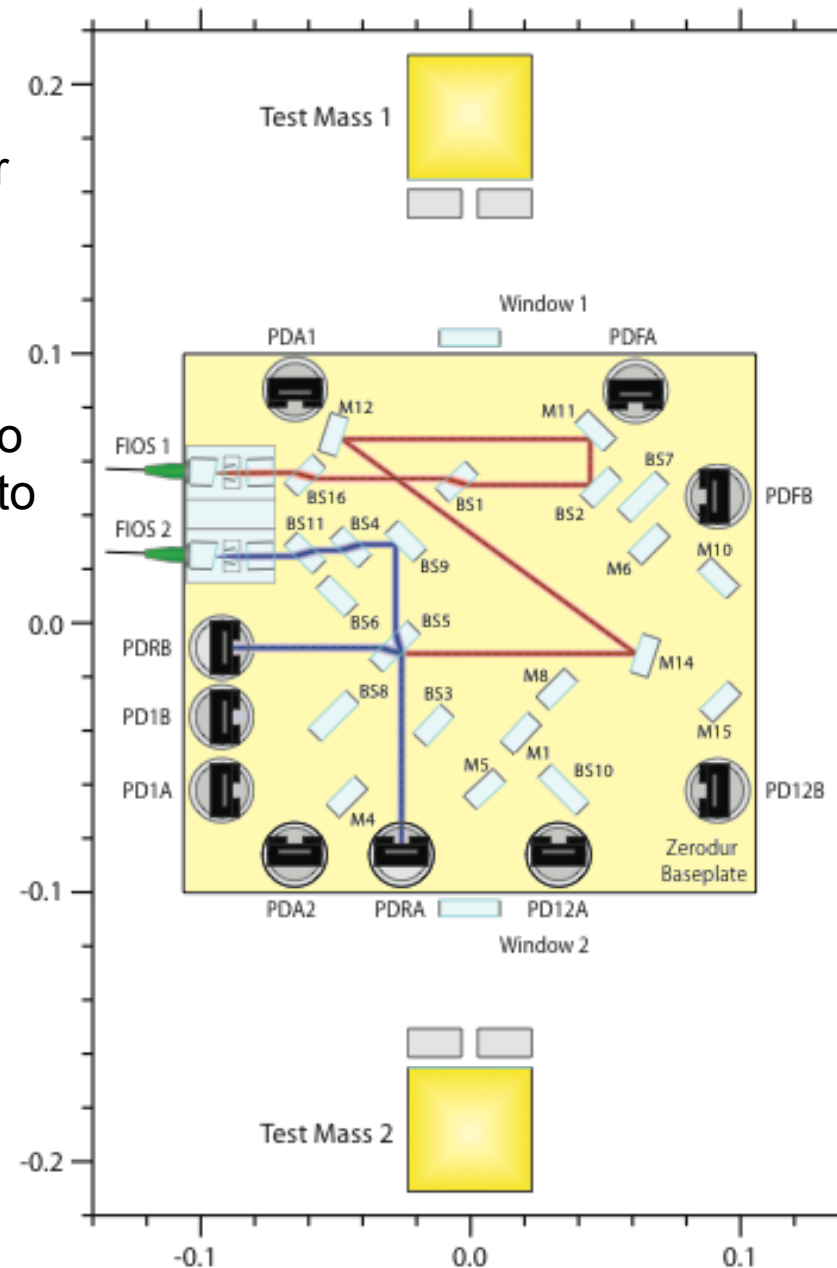
## X1 interferometer

- Relative displacement between test mass 1 and the spacecraft
  - $X_1$
- Relative angles between test mass 1 and the spacecraft
  - $\eta_1, \phi_1$



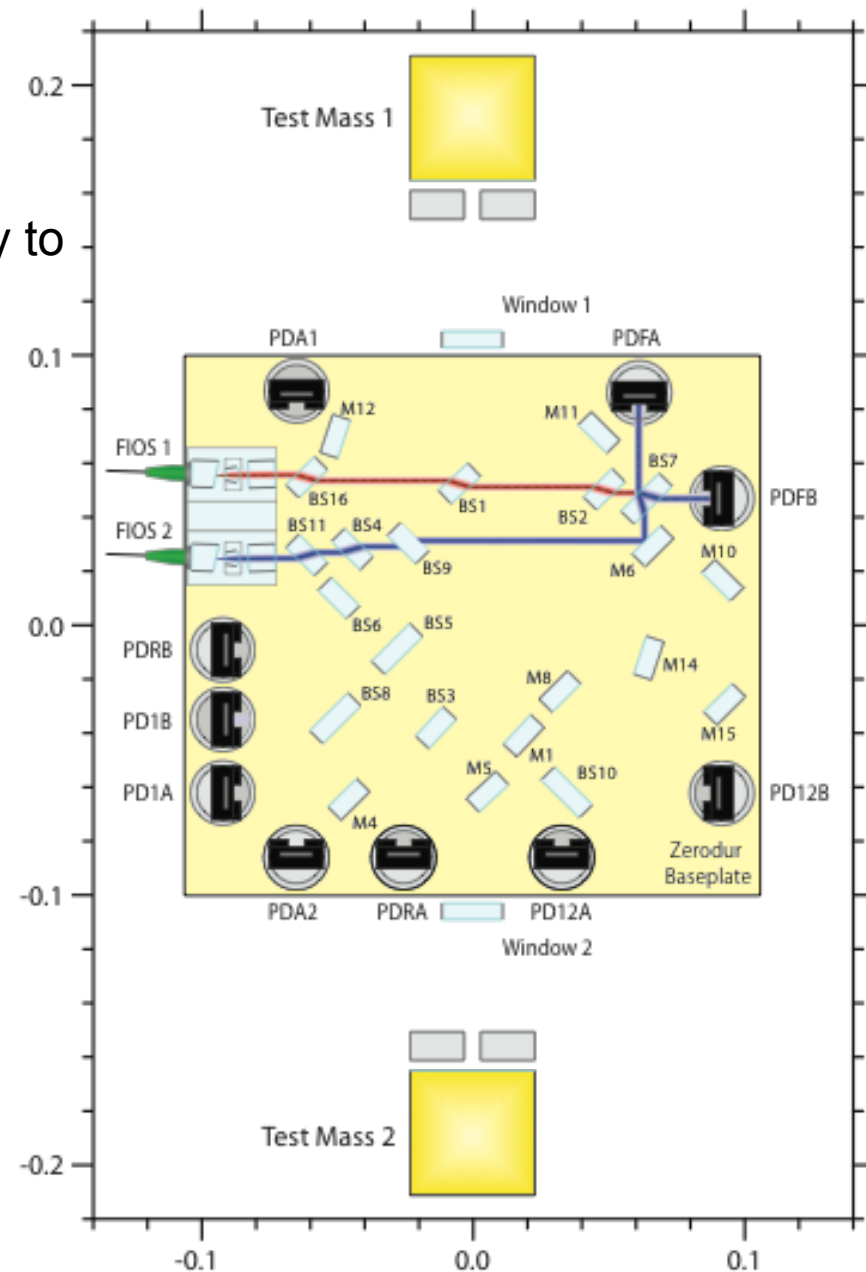
## Reference interferometer

- Provides reference phase against which other signals are compared
- Allows optical path length noise before the optical bench to be monitored
  - Relative path length noise between the two feed fibres can be controlled by feedback to actuators in the modulator unit (LM)



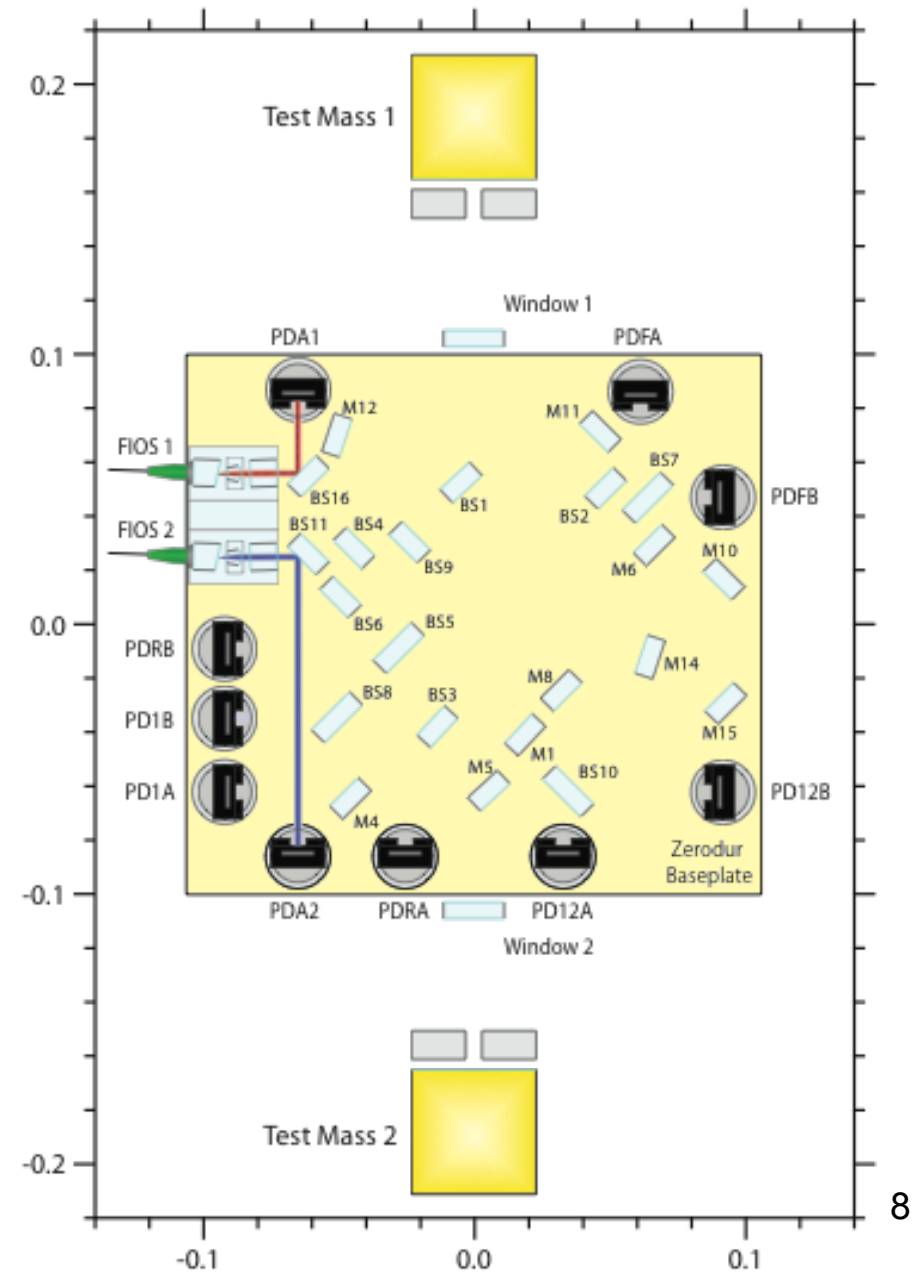
## Frequency noise interferometer

- Measure laser frequency noise
- Optical path unbalance to enhance sensitivity to laser frequency noise



## Power monitor

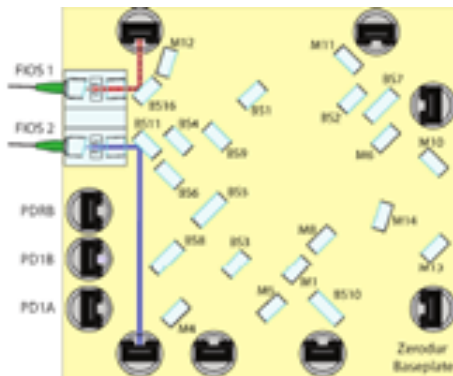
- Monitor the optical power from each fibre feed



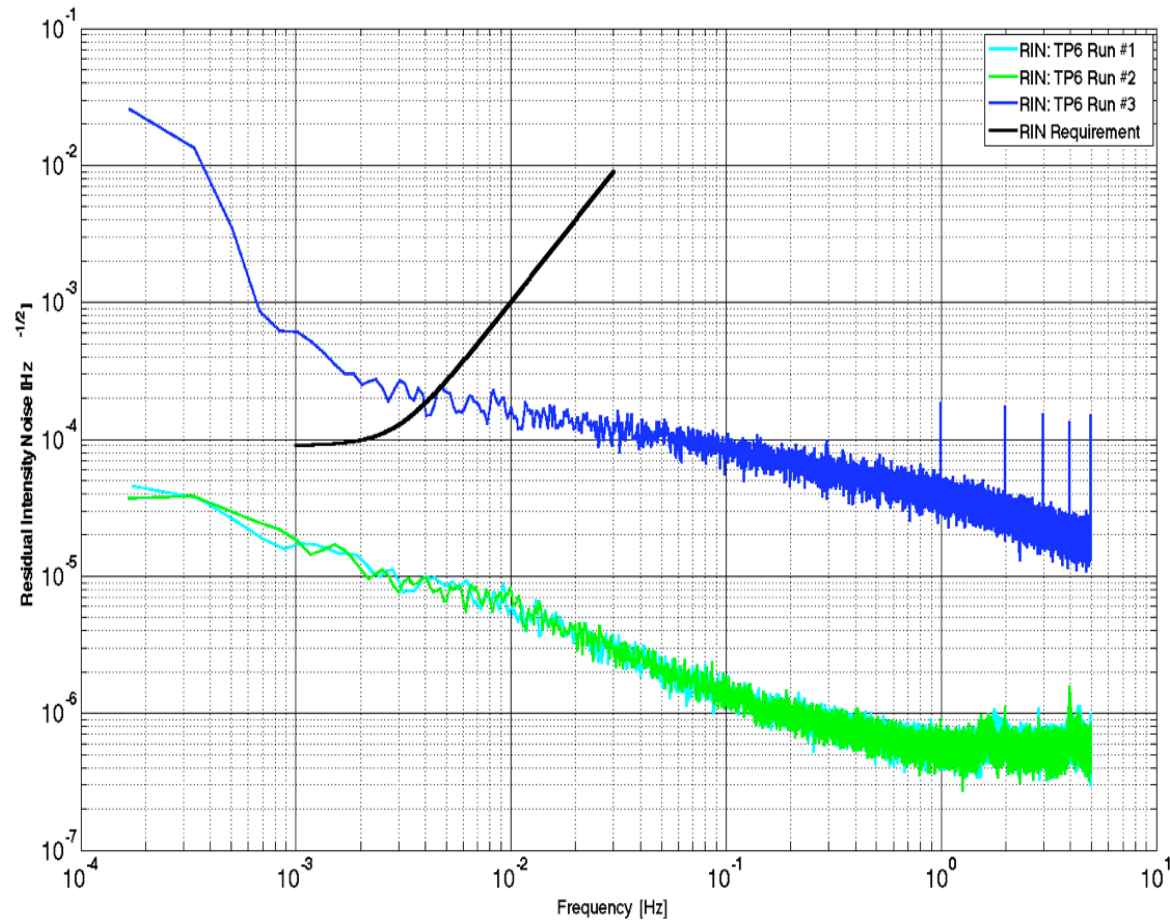


## Beam power noise

- Light beam onto test masses
- Direct force on test masses
- How to control laser power
  - Measure the noise
    - Allows subtraction from the data (OK)
    - Allows closed loop control (better)
  - Low noise laser (best)



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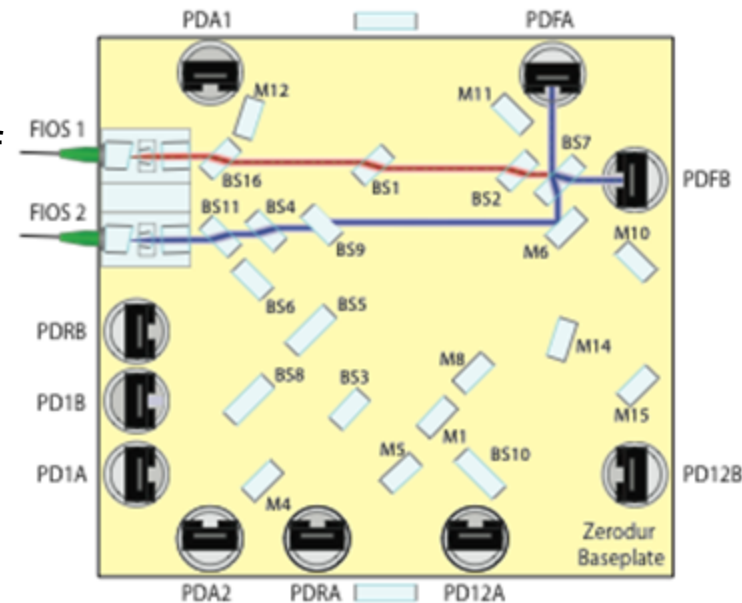


Residual laser power noise against frequency with the requirement in **black**– In the **blue** curve the control system is switched off

## Laser frequency noise

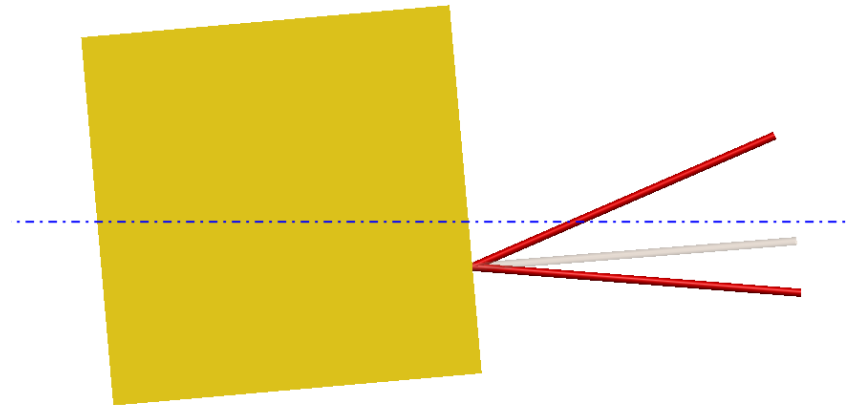
- Laser frequency noise  $\delta\nu$  can cause apparent test mass displacement noise  $\delta x$ 
  - Couples through path length mis-match  $\Delta L$
- Minimise effect by:
  - Well matched optical path lengths
    - Tracked during build matching which modelled the path length differences to be better than  $10\mu\text{m}$
    - Independent measurement gave upper limit of  $<100\mu\text{m}$  with a requirement of  $<1\text{mm}$
  - Measure f-noise with frequency noise interferometer
    - Path length mismatch 382mm
    - Control laser frequency or subtract
  - Low noise laser
- Results extremely good – see second last slide

$$\delta x = \Delta L \left[ \frac{\delta\nu}{\nu} \right]$$



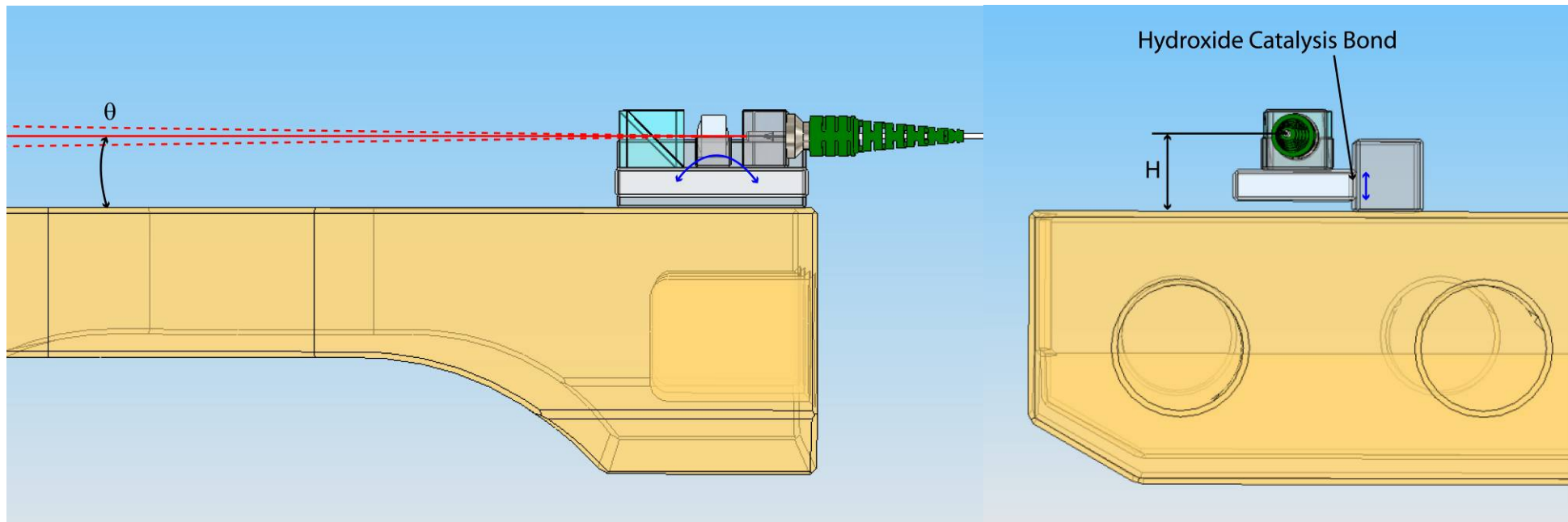
## Beam alignment onto test masses

- Why it matters
  - Geometrical piston effect
- OBI allowance is beam aligned to nominal point on test mass surface to better than  $\pm 25\mu\text{m}$
- How it is achieved
  
- Maintain alignment over thermal range
- Maintain alignment over launch and into operations
  
- Results



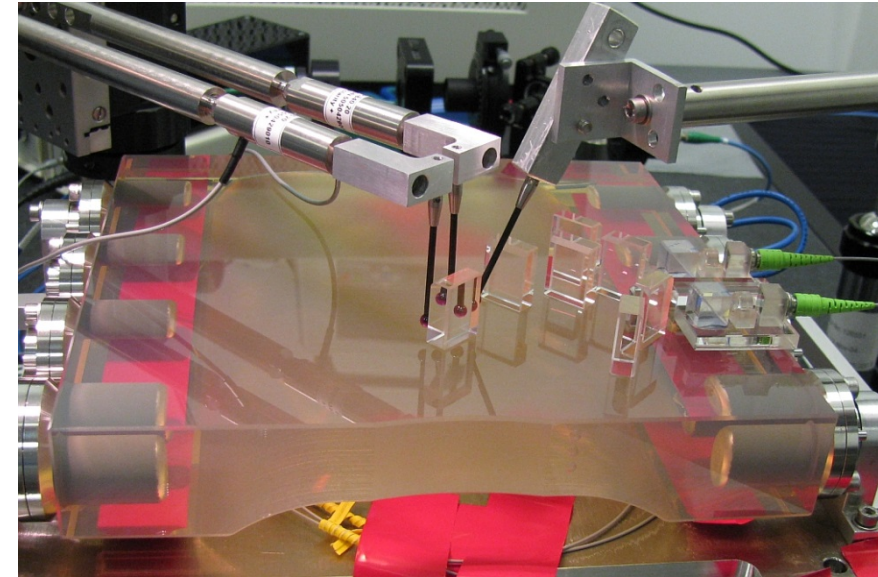
## Beam Alignment – out of plane

- Control the launch angle and height of beam launcher (FIOS)
- Target height at both test mass positions
  - Target error in  $H < 15\mu\text{m}$  (achieved  $10\mu\text{m}$ )
  - Target error in  $\theta < 30\mu\text{rad}$  (achieved  $24\mu\text{rad}$ )
- Subsequent components must be extremely perpendicular
  - 90 degrees to 1 arc second



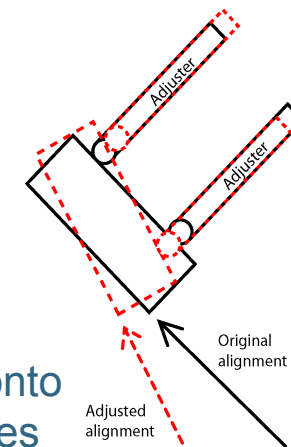
## Beam Alignment – in plane

- Control position and angle of critical components when bonding
  - Repeatability of positioning  $\sim 10\text{nm}$
  - Accuracy  $\sim 2\mu\text{m}$
- Critical components
  - Alignment onto test masses
  - Recombination beamsplitters
  - Optical path length matching

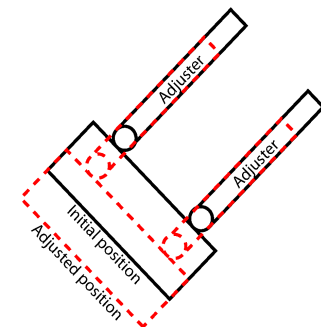


Flight Model	Y deviation ( $\mu\text{m}$ )	Z deviation ( $\mu\text{m}$ )
Test Mass1	-6	-15
Test Mass 2	-16	-7
Flight Spare	Y deviation ( $\mu\text{m}$ )	Z deviation ( $\mu\text{m}$ )
Test Mass1	15	-4
Test Mass 2	9	-1

Angular adjustment



Lateral adjustment

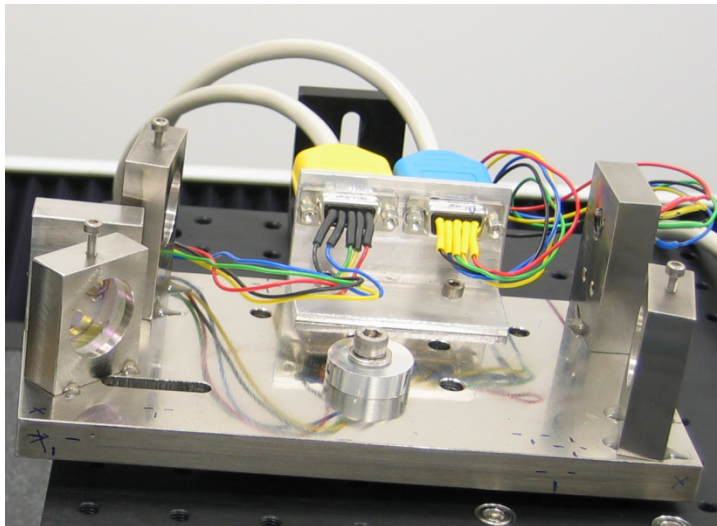
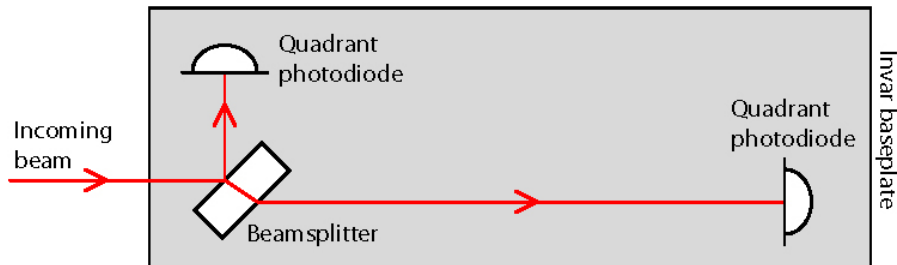


Alignment onto  
test masses

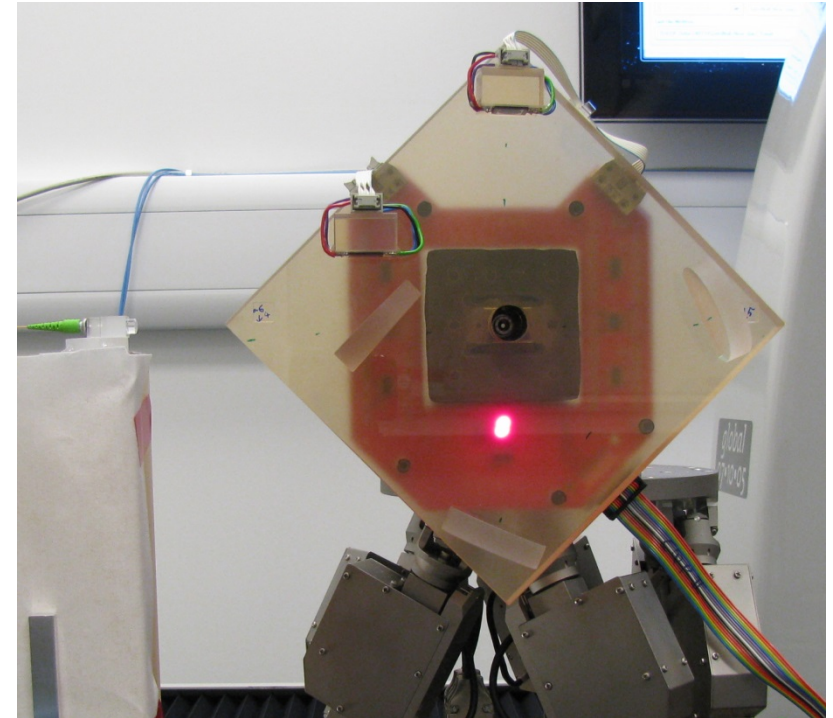
Target was  
 $< 25\mu\text{m}$

## Calibrated quadrant photodiode assembly (CQP)

- Absolute measurement of beam position
  - Combined with a Coordinate Measuring Machine
    - Physical measurement with accuracy  $<2\mu\text{m}$
    - Overall accuracy  $4\mu\text{m}$  and  $20\mu\text{rad}$



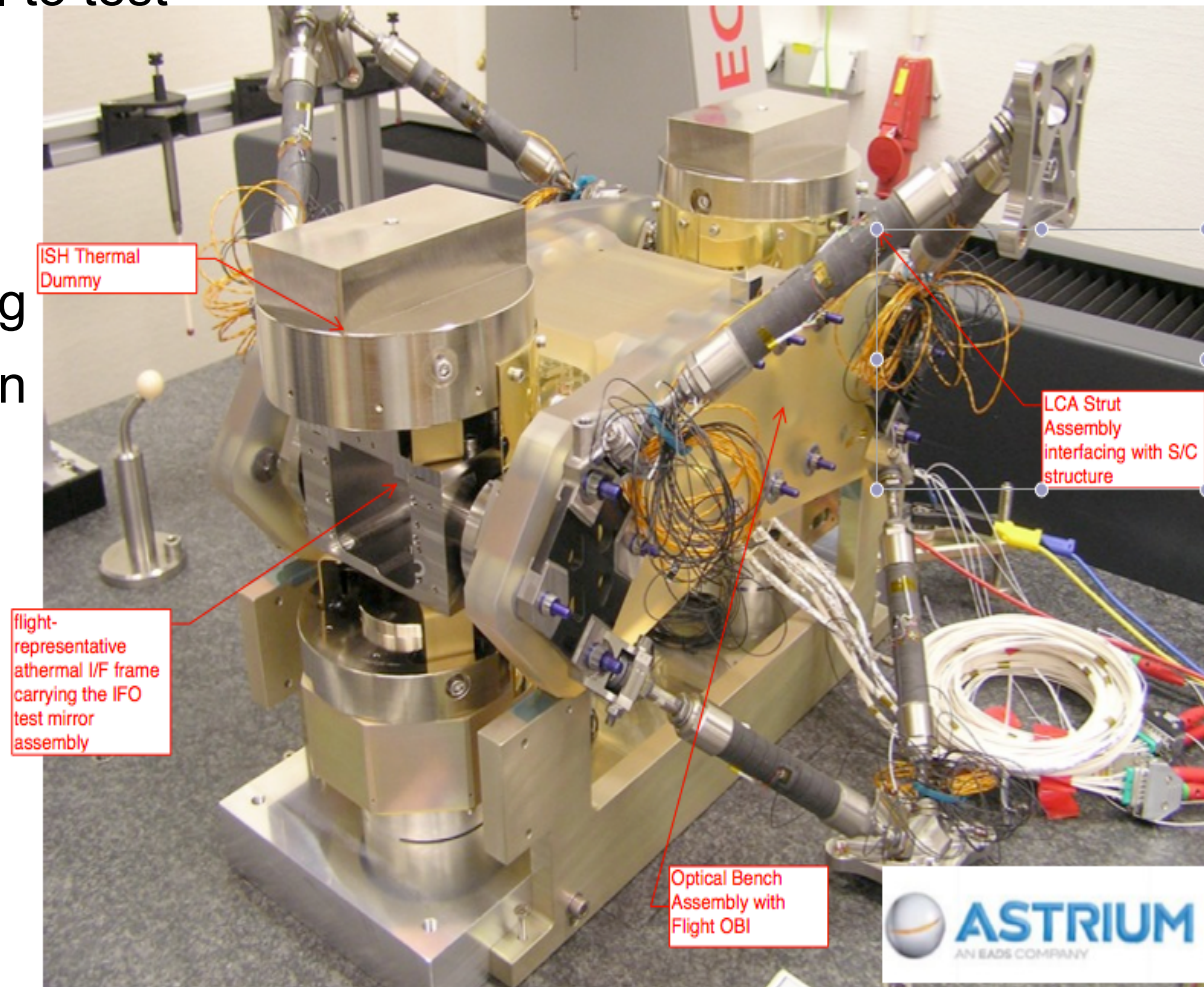
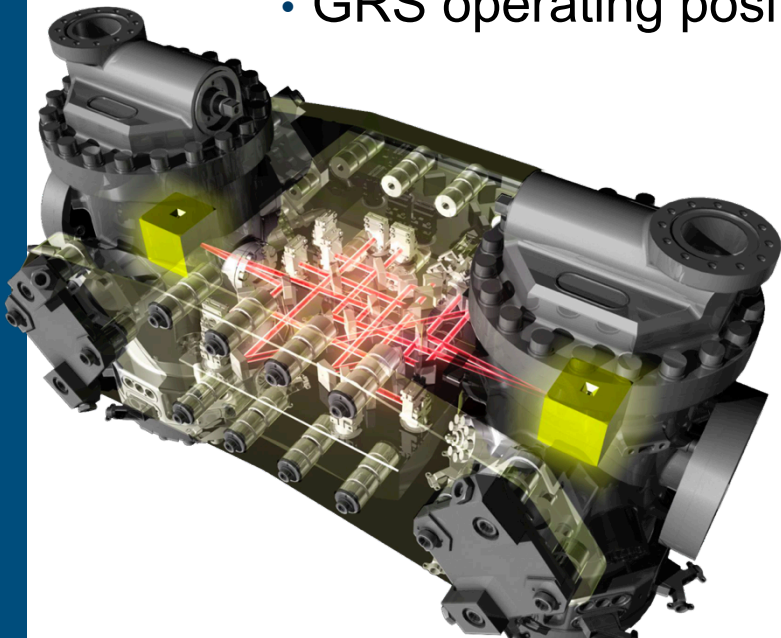
LPF  
CQP



Improved CQP

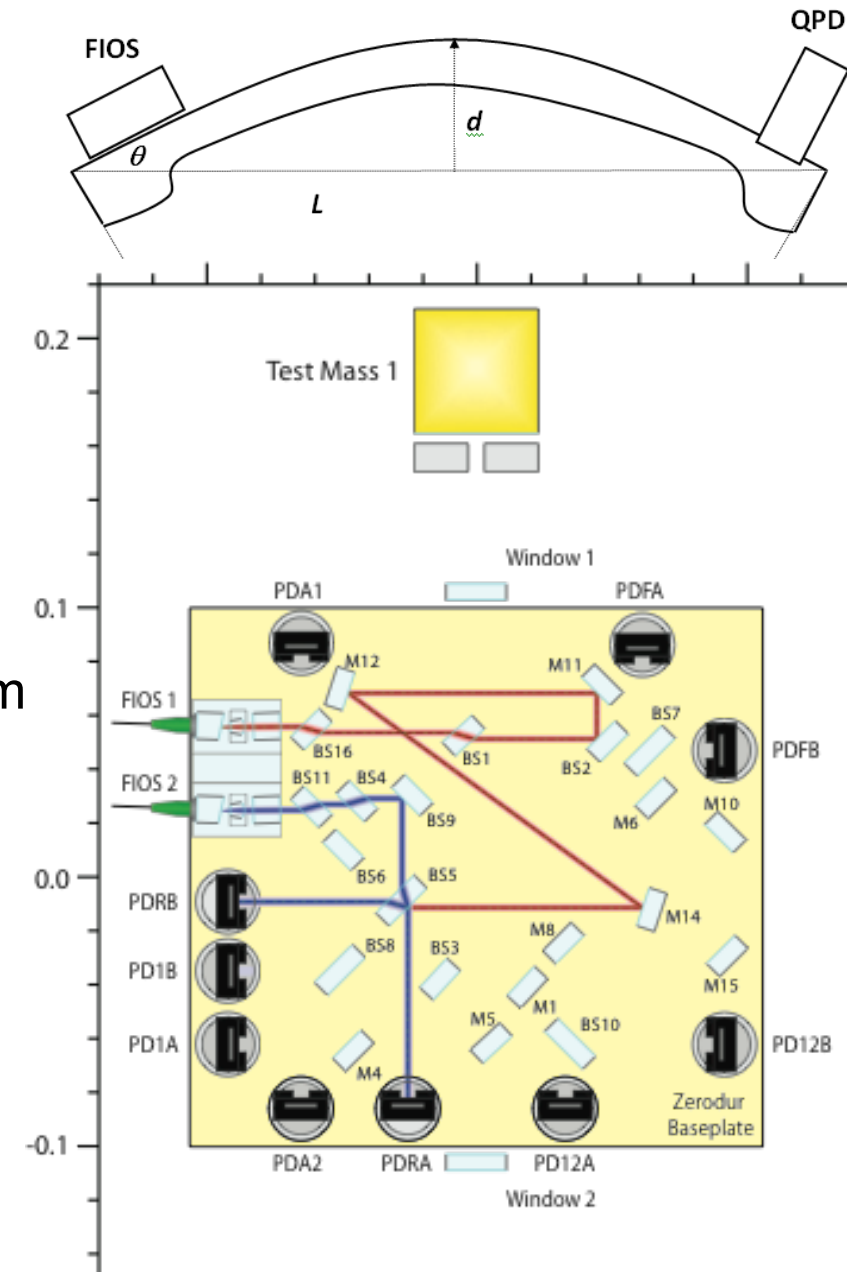
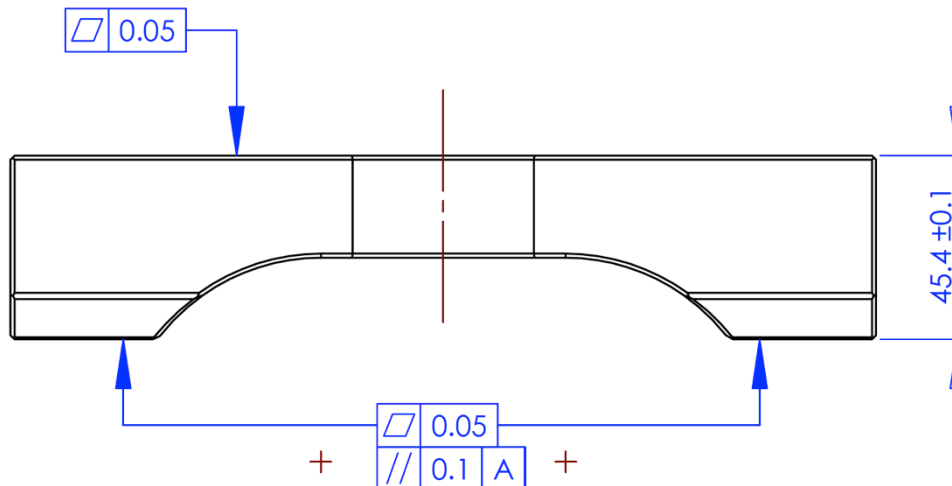
## OBI alignment is only part of the story

- Maintain alignment from OBI to test masses – alignment chain:
  - Side slabs
  - Vacuum can
  - GRS electrode housing
  - GRS operating position



## Mounting of the OBI must be stress free

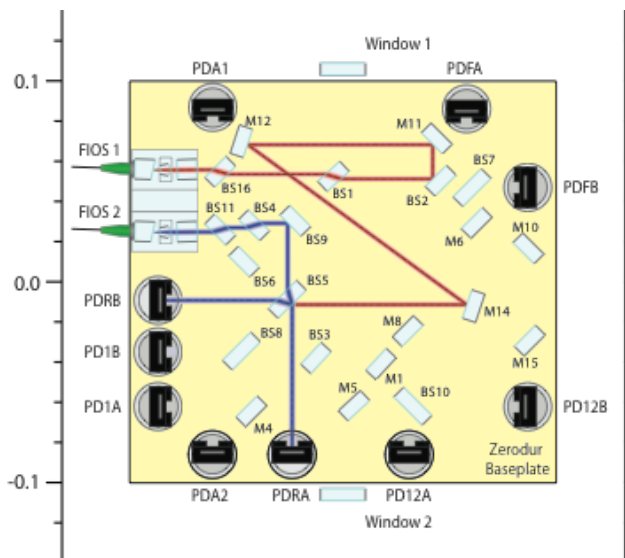
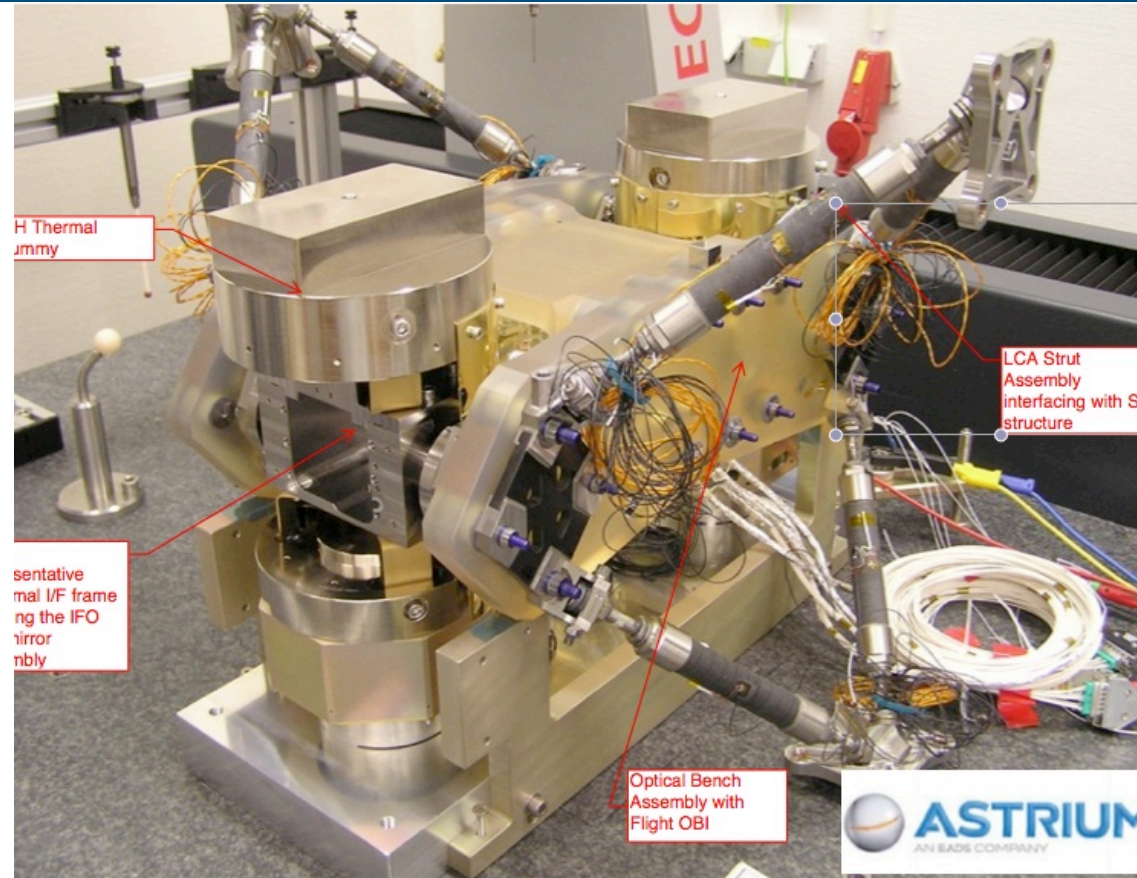
- Minimise mechanical stress in Zerodur
  - Structural safety
  - Minimise misalignments caused by distortion of the Zerodur OBI baseplate
    - These could cause misalignment of the beams onto the test masses
    - $1.7\mu\text{m}$  distortion ( $d$ ) gives  $37\mu\text{m}$  beam movement on photodiode PDRB





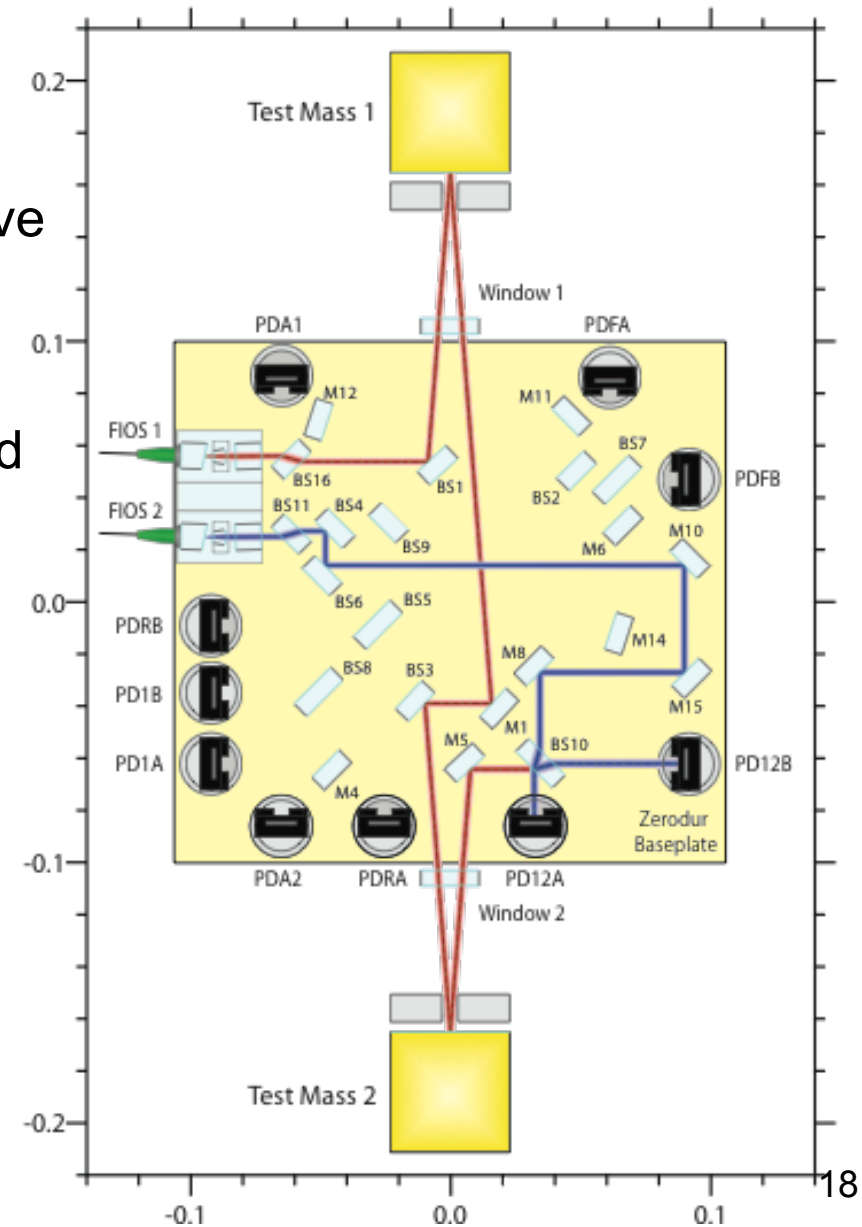
## Monitor beam positions on photodiodes during critical parts of the integration

- Minimising beam movements minimises OBI distortion
- System used during assembly



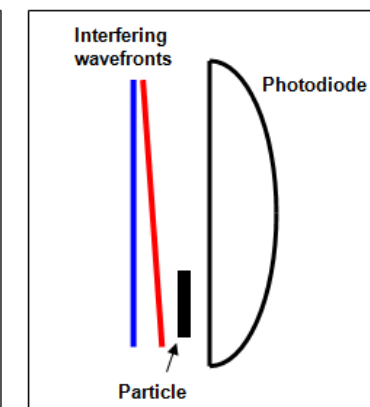
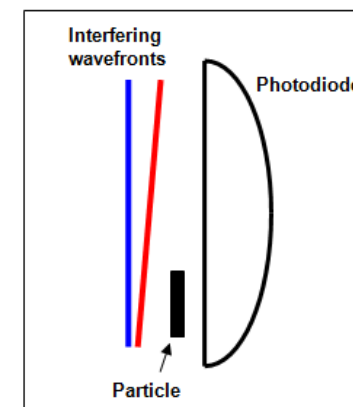
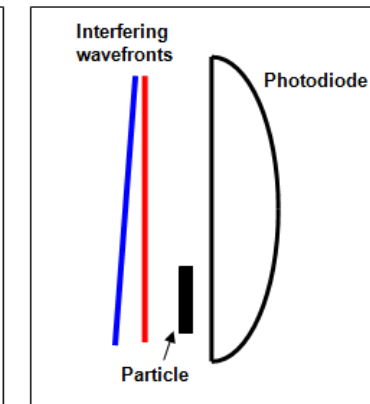
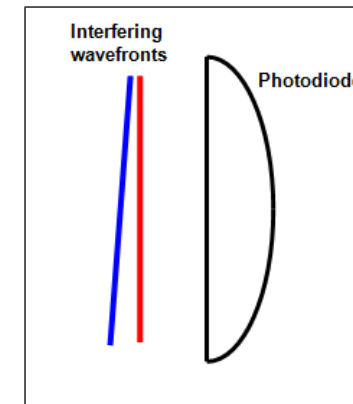
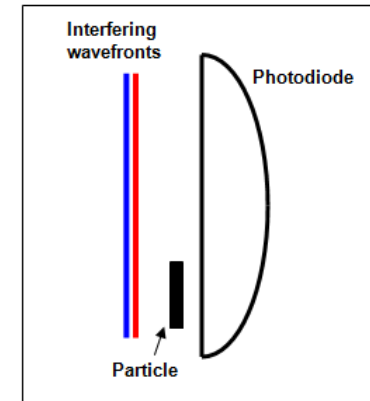
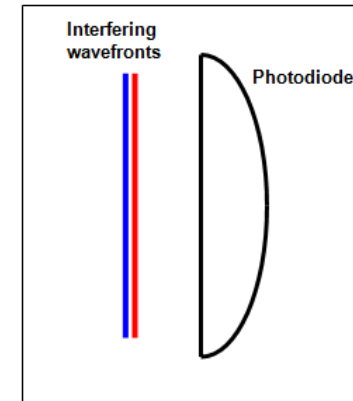
## OMS's main aim is to measure the freely floating test masses

- Optical beams reflected from them will move with respect to the OBI
- Potential noise couplings
  - Not noise performance tested on ground
  - Check expected performance through modelling and characterisation



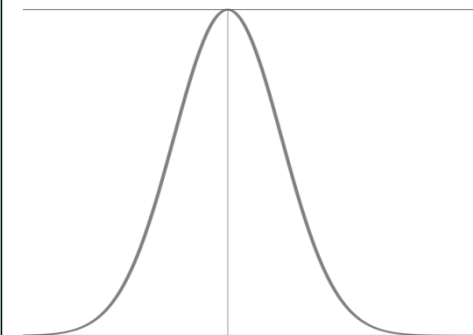
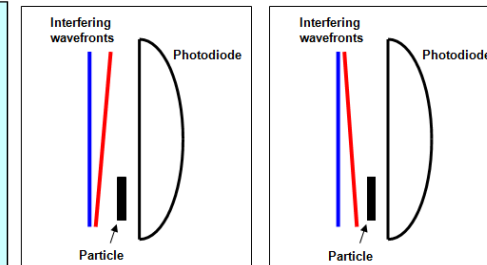
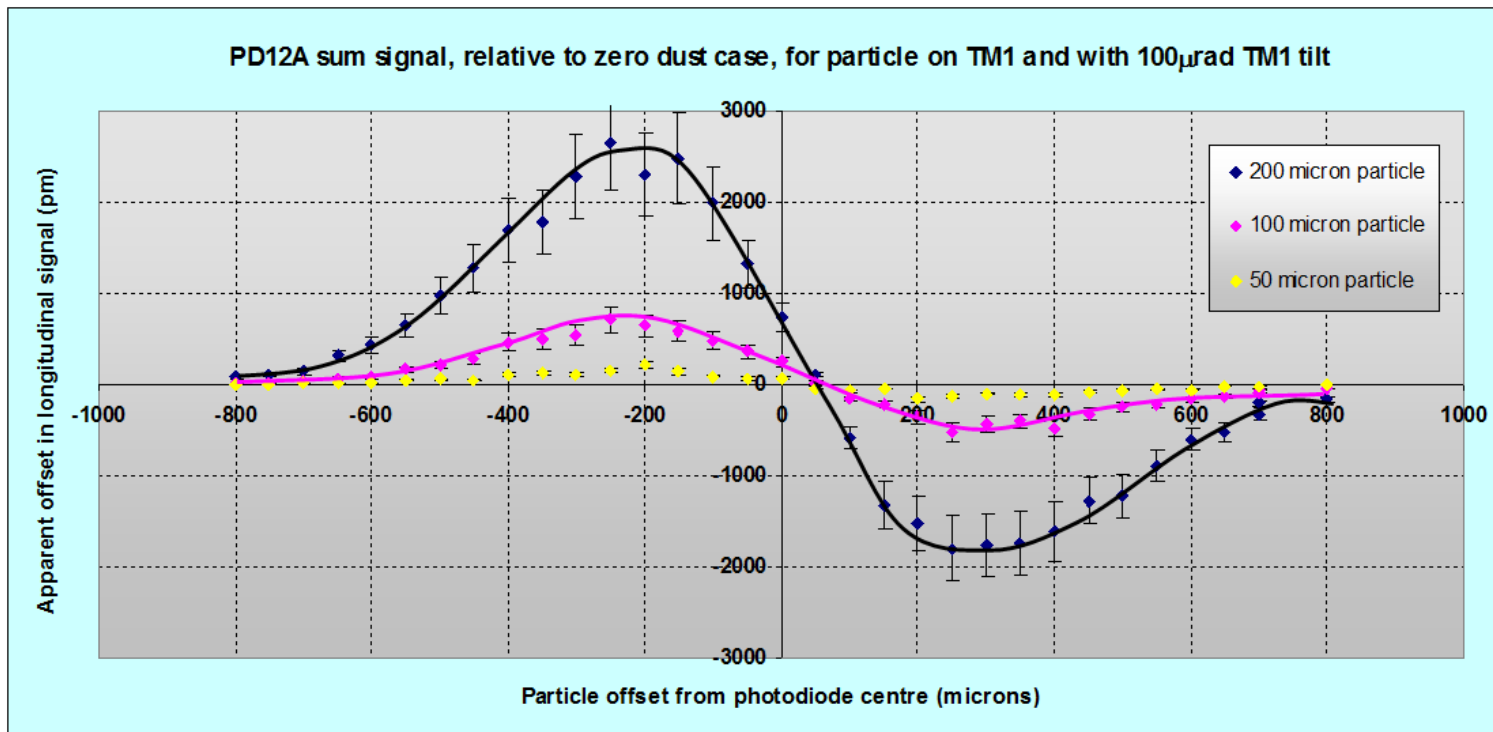
## Why beam obscuration matters

- Two interfering wavefronts
  - Diode detects the signal integrated over the whole beam
- 1) Parallel wavefronts with dust particle obscuring part of the beam
  - Signal amplitude reduced, no phase effect
- 2) Angled wavefronts
  - Signal amplitude reduced, static phase offset
- 3) Angled wavefronts
  - Signal amplitude reduced, phase offset varies with wavefront angle
  - Effectively a tilt to piston coupling due to the OMS



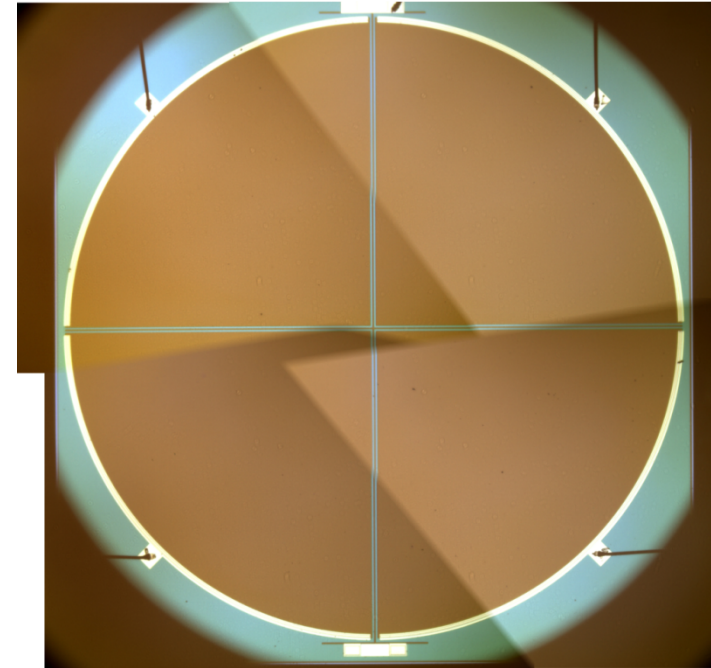
## Modelling

- Effect is zero if the contamination is centred on the interfering beams
- Maximum effect at a little under half a beam radius from beam centre
  - Set allowable contamination size of  $< 60\mu\text{m}$  for a particle in optimally bad position
    - $1\text{pm}/\sqrt{\text{Hz}}$  noise allowance, TM angular noise  $300\text{nrad}/\sqrt{\text{Hz}}$

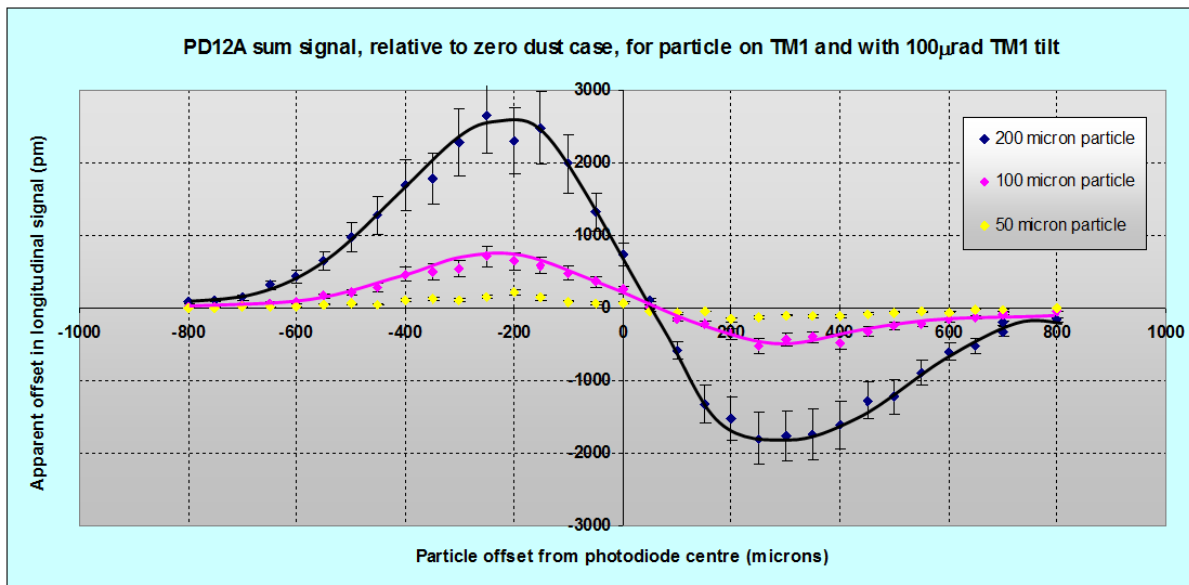


## Photodiode alignment

- Diameter 5mm
- Inter-quadrant gap  $45\ \mu\text{m}$ 
  - Acts like a linear dust particle
  - Photodiode must be centred on interfering beams to  $<33\ \mu\text{m}$
  - 1pm noise allowance, TM angular noise  $300\text{nrad}/\sqrt{\text{Hz}}$

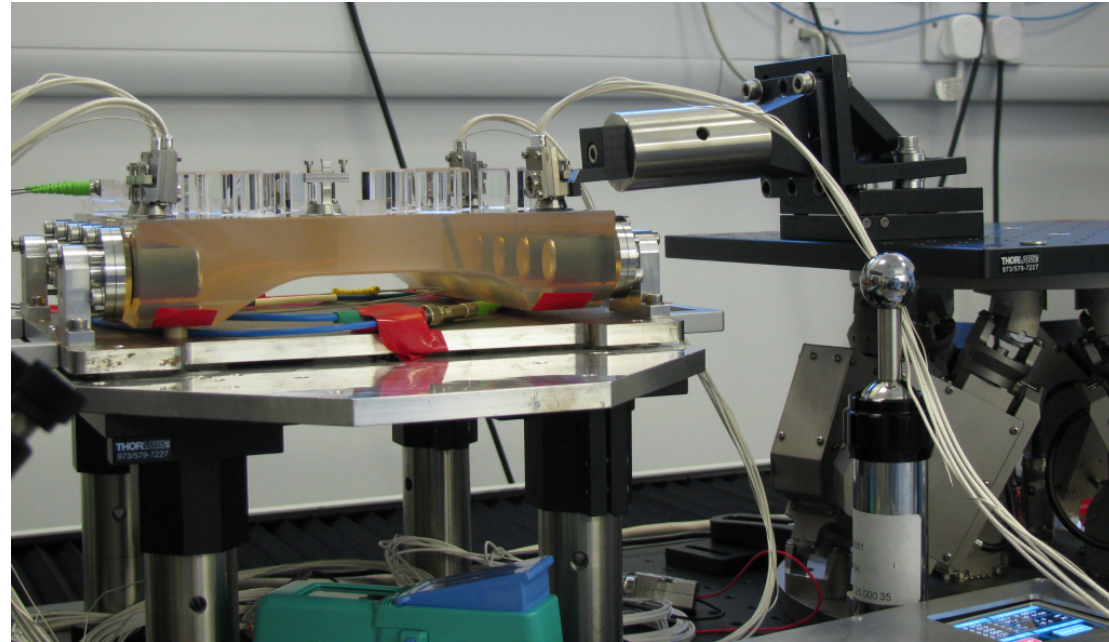


Composite image of an LTP InGaAs quadrant photodiode taken with an optical CMM



## LPF photodiode alignment

- All just about within specification on the moving beam interferometers



Aligning photodiodes onto the Flight Spare optical bench

Photodiode	Angle (degrees)
PD12A	-0.17
PD12B	-0.08
PD1A	-0.26
PD1B	-0.19
PDRA	0.05
PDRB	-0.02
PDFA	-0.22
PDFB	0.05

Photodiode  
angular  
rotation

Target <0.5  
degrees

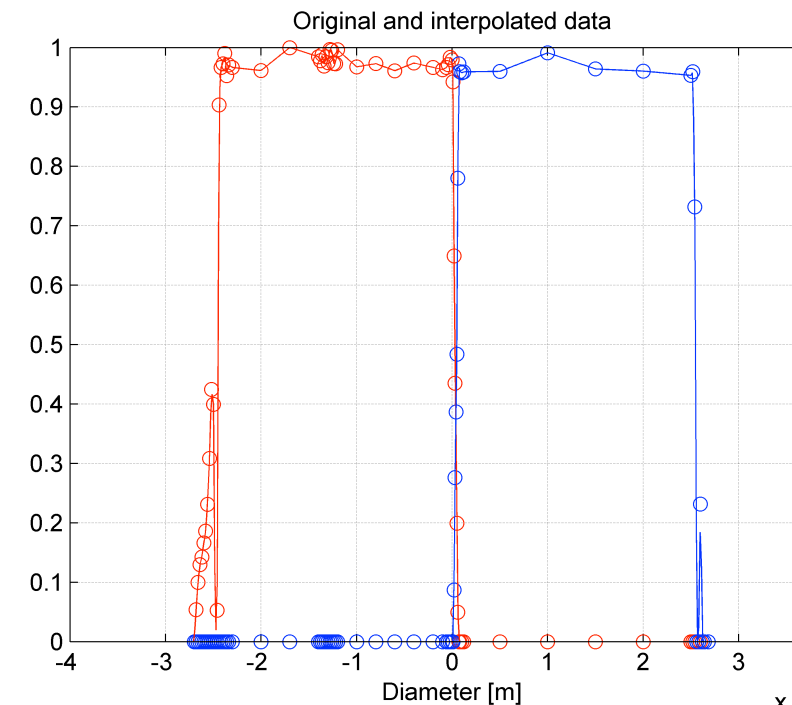
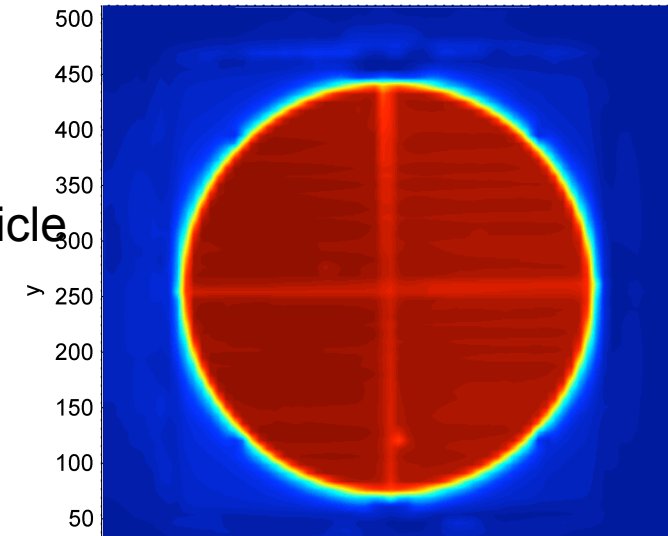
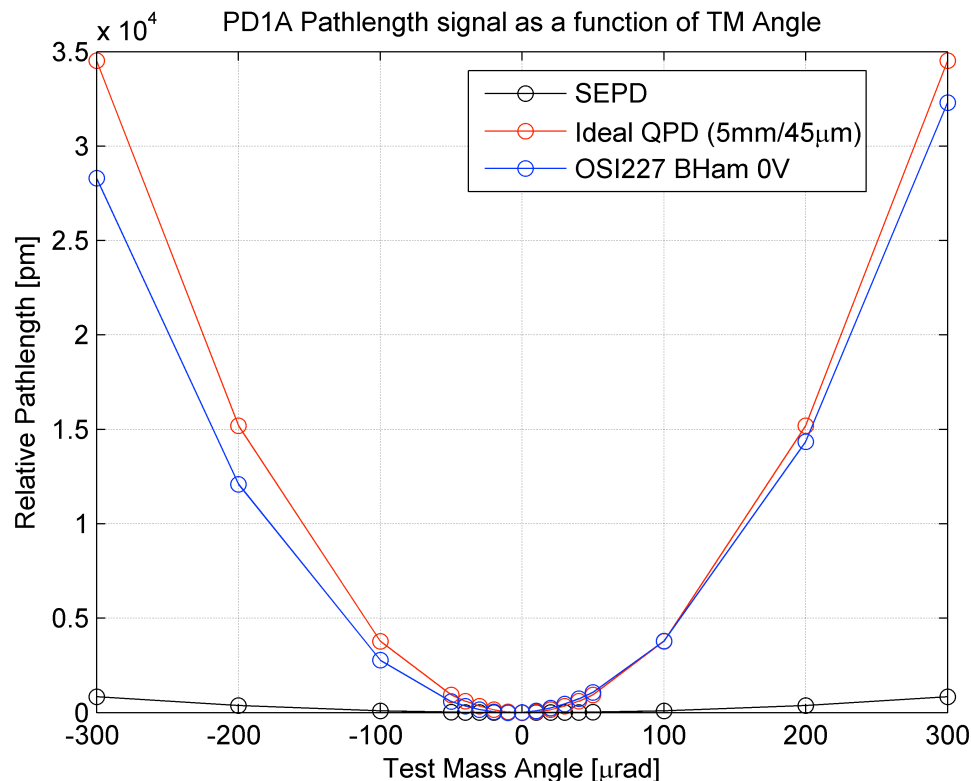
	X(mm)	Y(mm)
PD12A	8	0
PD12B	12	37
PD1A	12	3
PD1B	-5	-5
PDRA	-13	-6
PDRB	58	89
PDFA	21	-29
PDFB	8	-13

Moving beam  
interferometers

Fixed beam  
interferometers

## Flight QPDs are InGaAs

- Uniformity of response measured by scanning beam
- Effect of non-uniformity is similar to a diffuse dust particle
- Flight model QPDs very uniform response
- Very sharp division between quadrants
- Tilt to piston marginally better than “ideal” QPD



## On Station Thermal Tests

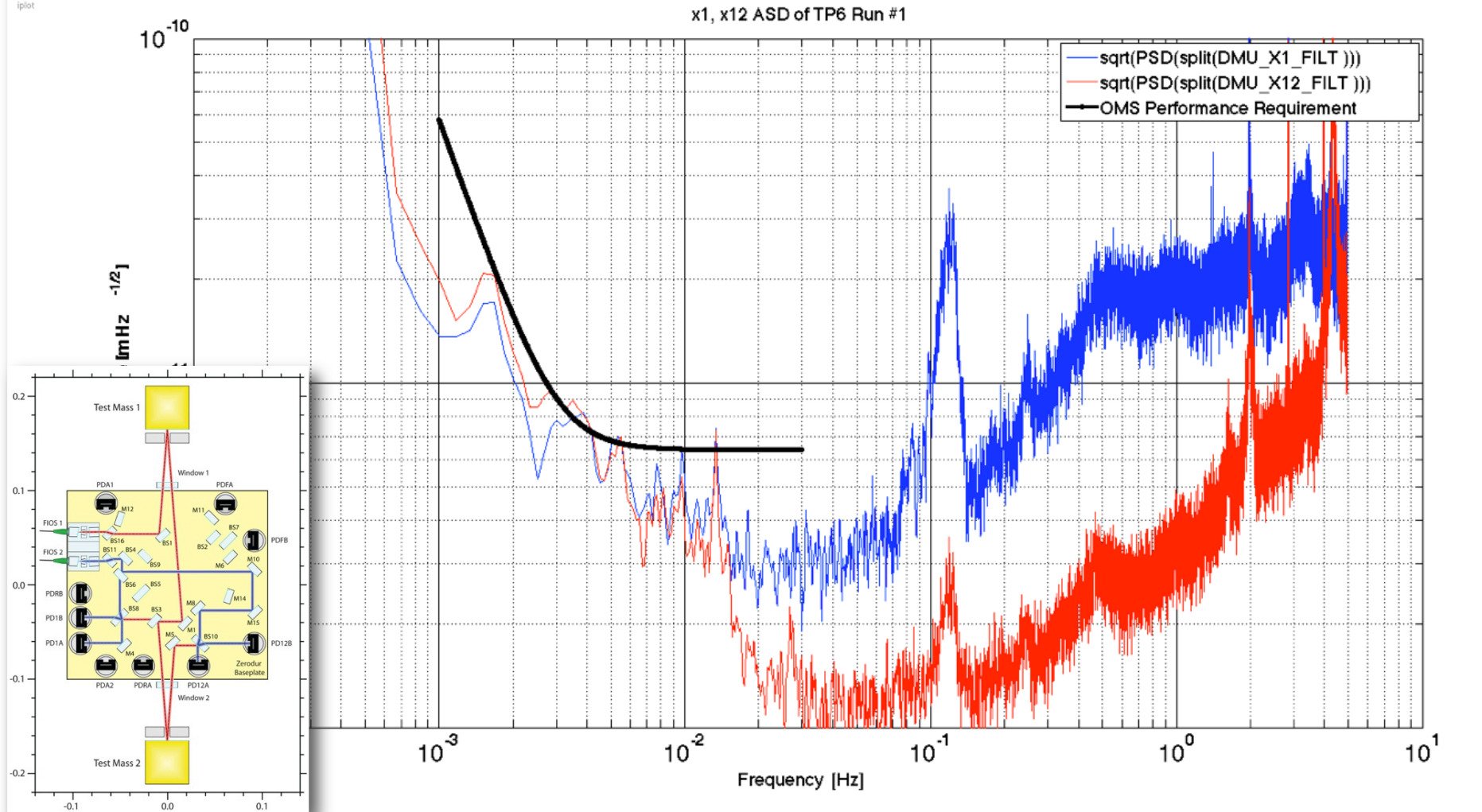
- Full optical metrology system
- Static dummy mirrors in place of test masses
- Overall performance exceeded expectations



LPF Spacecraft being prepared for OSTT campaign



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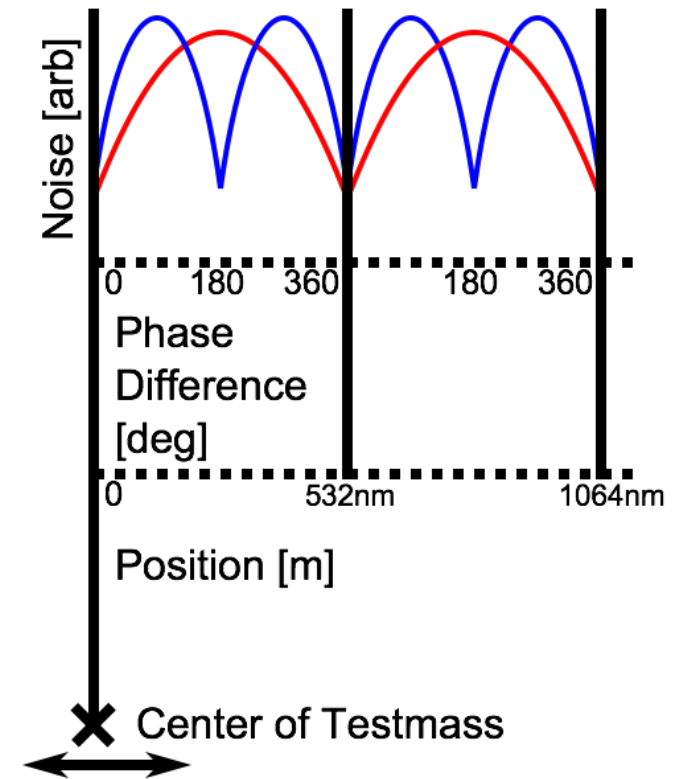


## Out of band noise can couple to OMS measurements

- Amplitude noise
- Phase noise
- Coupling can be minimised by moving test masses longitudinally to an optimum operating point
  - Test mass movements of  $<\lambda$
- Effect seen in OSTT data
- See **“Optimising Test Mass Position for the LISA-Pathfinder Optical Metrology System”** by Andreas Wittchen on Thursday afternoon

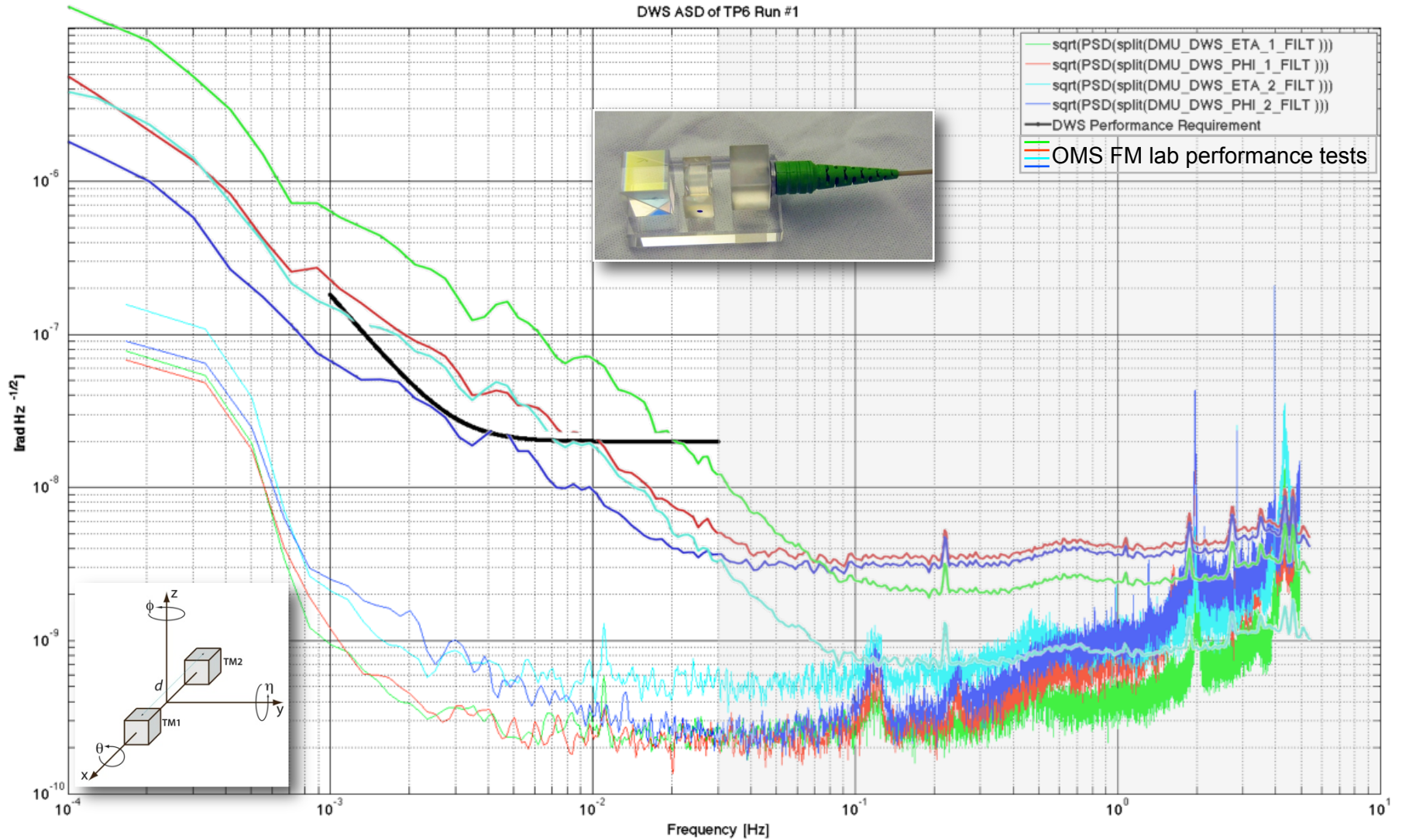
amplitude noise

phase noise



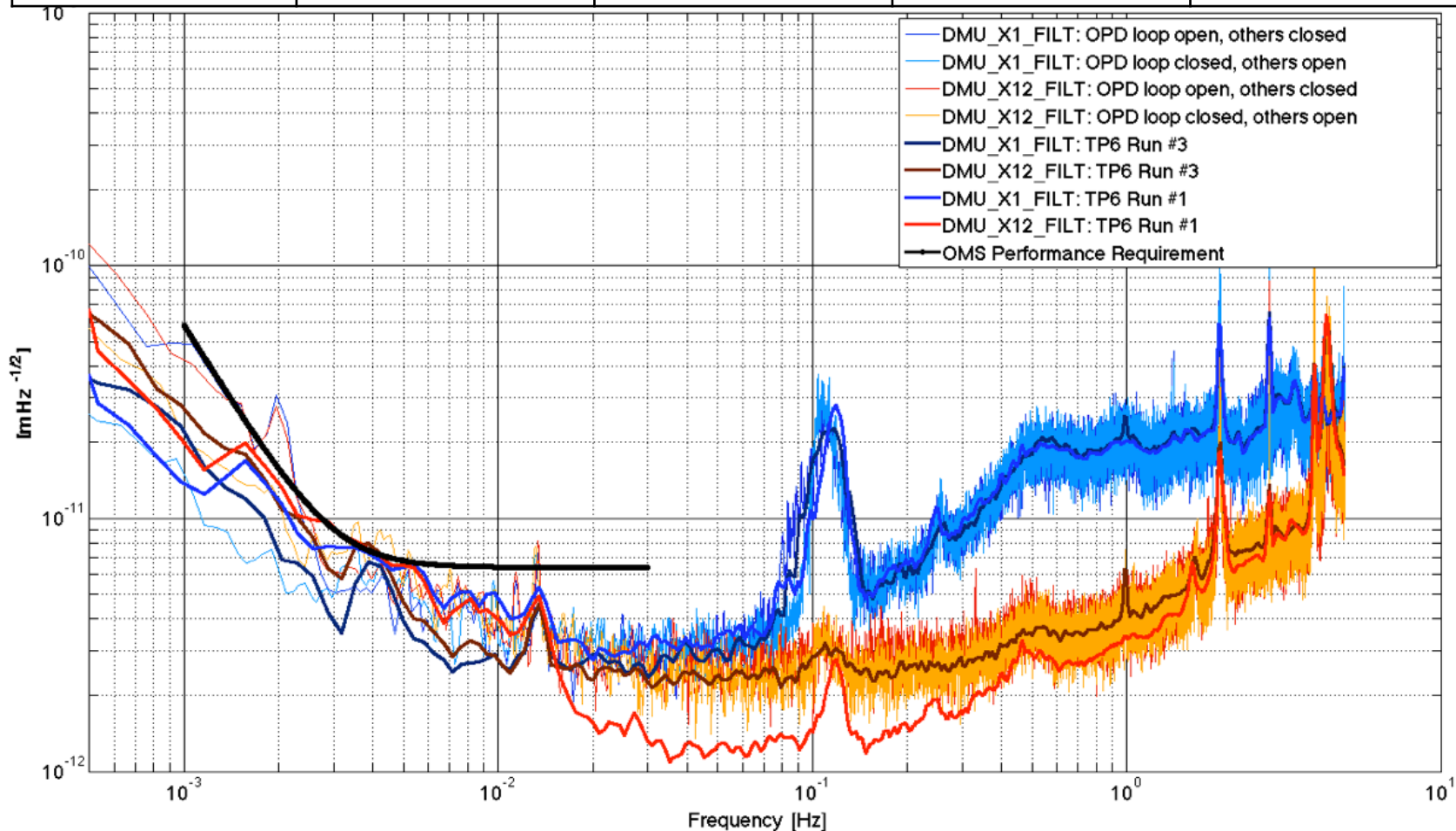
Noise coupling with test Mass position - See Andreas Wittchen's presentation for full details

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gslx



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iplot

Run Number	Power Loop	Fast Frequency	Slow Frequency	OPD Loop
1	Yes	Yes	Yes	No
2	Yes	No	No	Yes
TP6 Run #3	No	Yes	Yes	Yes
TP6 Run #1	Yes	Yes	Yes	Yes



## Conclusions

- LTP OMS technology fully integrated into LPT core assembly
- Building the LTP OMS has exercised many components, technologies and systems relevant to LISA local interferometry
- Testing at component and system level ~~very satisfactory~~ **AWESOME!**
  - Static noise performance
  - Noise coupling measurements
  - Performance meets or exceeds goals
  - Some interesting features
    - Strategies to deal with them in place
- Now needs tested on orbit!