

10/14/21

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Announcements

HW 4 due 10/21

Quiz 4 Monday 10/17 Newton's laws in rot. frame, Coriolis force

Last time: tides

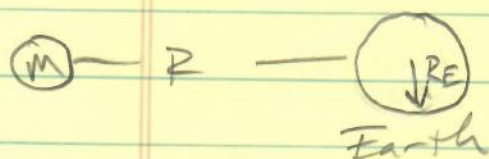
We found just from applying $F = -\frac{GM_1M_2}{r^2}$ to Earth + oceans

effective potential near Earth $r \ll R$

$$\Phi = \frac{U}{m} = -\frac{GM_e}{r} - \frac{GM_m r^2}{2R^3} (3\sin^2\theta \cos^2\phi - 1)$$

Water surface is equipotential

$$\Rightarrow h(\theta, \phi) = r - R_e = \frac{R_e}{2} \left(\frac{M_m}{M_e} \right) \left(\frac{R_e}{R} \right)^3 (\sin^2\theta \cos^2\phi - 1)$$



$$\Delta h = \frac{3R_e}{2} \left(\frac{M_m}{M_e} \right) \left(\frac{R_e}{R} \right)^3 \sin^2\theta$$

$$\approx 0.56 \text{ m}$$

max average

- tides max at equator $\theta = 90^\circ$, 0 at poles
(Not quite true because of Earth's tilt)

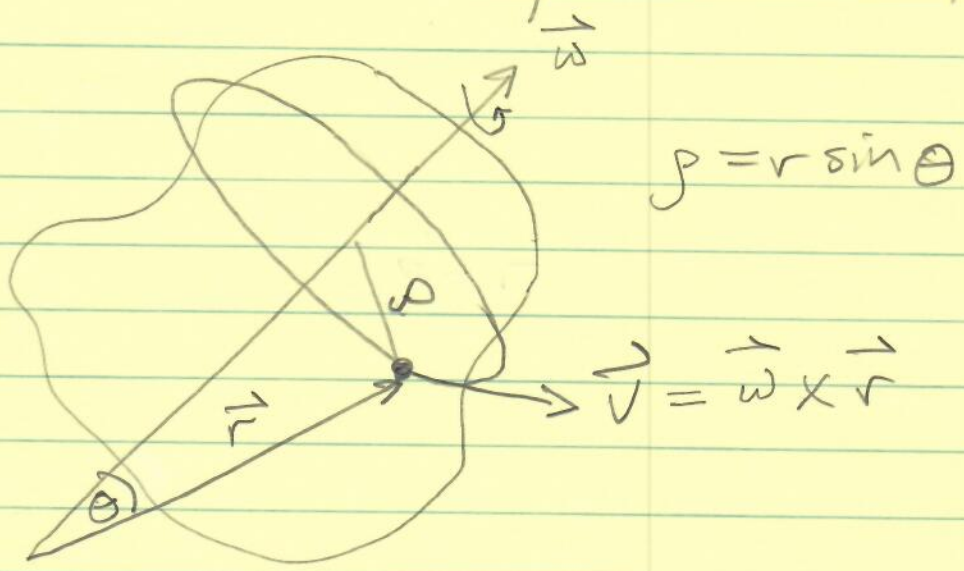
- Note our calculation applies also to Earth-sun system! $\Delta h \approx \frac{3R_e}{2} \left(\frac{M_s}{M_e} \right) \left(\frac{R_e}{R_{e-s}} \right)^3 \sin^2\theta$
Sun factor 2 weaker
not negligible! $\approx 0.25 \text{ m}$

Newton's laws in rotating frame
e.g. Earth!

Angular velocity vector - $\vec{\omega} = \omega \hat{\omega}$
book calls it $\vec{\omega} = \omega \hat{u}$ ↑
unit vector

direction of $\vec{\omega}$ determined by right hand rule.

- relation between rotating rigid body $\vec{\omega}$
and linear velocity \vec{v} of some point!



* This is for the vector \vec{r} $\frac{d\vec{r}}{dt} = \vec{\omega} \times \vec{r}$
but it holds for any vector
fixed in the frame of the body!

e.g. \hat{e} is rotating vector in body

$$\left. \frac{d\hat{e}}{dt} \right|_{so} = \vec{\omega} \times \hat{e}$$

↑
calculated in inertial frame

— addition of angular velocities

Frames S_0 - inertial and S - rotating

S_0 S_0 and S have relative angular velocity $\vec{\Omega}$

Recall: if two frames $Z, 1$ have relative velocity \vec{v}_{21} , and body 3 has rel. vel. \vec{v}_{32} relative to Z ,

$$\begin{aligned} \vec{v}_{31} &= \vec{v}_{32} + \vec{v}_{21} \\ \Rightarrow \vec{\omega}_{31} \times \vec{r} &= \vec{\omega}_{32} \times \vec{r} + \vec{\omega}_{21} \times \vec{r} \\ &= \vec{\omega}_{32} \times \vec{r} + \vec{\Omega} \times \vec{r} \end{aligned}$$

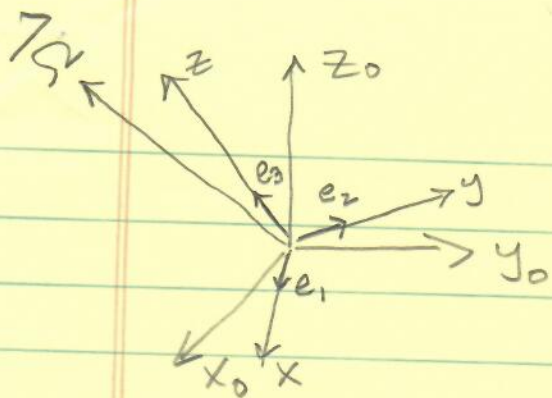
True $\forall \vec{r} \Rightarrow$ $\vec{\omega}_{31} = \vec{\omega}_{32} + \vec{\Omega}$

— Time derivatives in rotating frame

Consider arbitrary vector \vec{Q} varying in time

Question: how are $\left. \frac{d\vec{Q}}{dt} \right|_{S_0}$ and $\left. \frac{d\vec{Q}}{dt} \right|_S$ related?

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note \hat{z} doesn't necessarily point along $\vec{\Omega}$

Express \vec{Q} in terms of unit vectors $\hat{e}_1, \hat{e}_2, \hat{e}_3$ fixed in S (not S₀)

$$\vec{Q} = Q_1 \hat{e}_1 + Q_2 \hat{e}_2 + Q_3 \hat{e}_3 = \sum_i Q_i \hat{e}_i$$

In S, \hat{e}_i are constant in time
Therefore

$$\left. \frac{dQ}{dt} \right|_S = \sum_i \frac{dQ_i}{dt} \hat{e}_i$$

★ Vector \vec{Q} same in both frames \Rightarrow components Q_i also same. An observer in S₀ sees \hat{e}_i rotating with time, but says Q_i measured with respect to these directions are the same.

$\Rightarrow \frac{dQ_i}{dt}$ is same for both S + S₀

In S₀ $\hat{e}_i = \hat{e}_i(t)$, so

$$\left. \frac{d\vec{Q}}{dt} \right|_{S_0} = \sum_i \frac{dQ_i}{dt} \hat{e}_i + \sum_i Q_i \left. \frac{d\hat{e}_i}{dt} \right|_{S_0}$$

but we said for any vector \vec{e} fixed in S ,

$$\left. \frac{d\vec{e}}{dt} \right|_{S_0} = \vec{\omega} \times \vec{e}$$

$$\text{so } \left[\begin{aligned} \left. \frac{d\vec{Q}}{dt} \right|_{S_0} &= \sum_i \frac{dQ_i}{dt} \hat{e}_i + \sum_i Q_i \vec{\Omega} \times \hat{e}_i \\ &= \left. \frac{d\vec{Q}}{dt} \right|_S + \vec{\Omega} \times \vec{Q} \end{aligned} \right]$$

Apply now to Newton's laws:

In inertial frame we know $m \left. \frac{d^2 \vec{r}}{dt^2} \right|_{S_0} = \vec{F}$

In rotating frame,

$$\left. \frac{d\vec{r}}{dt} \right|_{S_0} = \left. \frac{d\vec{r}}{dt} \right|_S + \vec{\Omega} \times \vec{r}$$

$$\frac{d}{dt} \text{ again: } \left. \frac{d^2 \vec{r}}{dt^2} \right|_{S_0} = \left. \frac{d}{dt} \right|_{S_0} \left(\left. \frac{d\vec{r}}{dt} \right|_{S_0} \right)$$

$$= \left. \frac{d}{dt} \right|_{S_0} \left[\left. \frac{d\vec{r}}{dt} \right|_S + \vec{\Omega} \times \vec{r} \right]$$

$$= \left. \frac{d}{dt} \right|_S [] + \vec{\Omega} \times []$$

* Notation: in this context, use dot \dot{Q} to mean $\left. \frac{dQ}{dt} \right|_S$

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$$\Rightarrow \left. \frac{d^2 \vec{r}}{dt^2} \right|_{S_0} = \ddot{\vec{r}} + \vec{\Omega} \times \dot{\vec{r}} + \dot{\vec{\Omega}} \times \vec{r} + \dot{\vec{\Omega}} \times (\vec{\Omega} \times \vec{r})$$

$$m \left. \frac{d^2 \vec{r}}{dt^2} \right|_{S_0} = m \ddot{\vec{r}} + m 2 \vec{\Omega} \times \dot{\vec{r}} + m \dot{\vec{\Omega}} \times (\vec{\Omega} \times \vec{r})$$

$$\underbrace{\quad}_{\vec{F}}$$

Coriolis
Force

Centrifugal
Force

$$m \ddot{\vec{r}} = \vec{F} + 2m \dot{\vec{r}} \times \vec{\Omega} + m (\dot{\vec{\Omega}} \times \vec{r}) \times \vec{\Omega}$$

rotating
frame

inertial
frame